AIRCRAFT SPECIFICATIONS

MODEL:	B737-205 Advanced	MAX. TAXI WEIGHT:	53,297	kg	117,500	lbs
MANUFACTURER:	Boeing	MAX. GROSS TAKE-OFF WT.:	53,070	kg	117,000	lbs
SERIAL NUMBER:	21445	MAX. LANDING WEIGHT:	46,720	kg	103,000	lbs
LINE / VARIABLE NUMBER:	506 / PJ006	MAX. ZERO FUEL WEIGHT:	43,091	kg	95,000	lbs
REG. NUMBER:	CC-CYK	MFGRS. EMPTY WEIGHT:	---	kg	---	lbs
DATE OF MANUFACTURE:	December 3, 1977	OPERATORS WEIGHT EMPTY:	29,020	kg	63,978	lbs
CURRENT OPERATOR:	LAN Express	PAYLOAD:	14,071	kg	31,022	lbs

AIRFRAME STATUS

TIME AS OF:	June 21, 2006	MTX. PROGRAM:	LAN Express:
TOTAL AIRFRAME HOURS:	65,721	A Check every 150 hours	
TOTAL AIRFRAME CYCLES:	76,639	B Check every 750 hours	
TIME SINCE "1C" CHECK:	1,440 Hours (May 2005)	C Check and multiples every 3,500 hours	
TIME SINCE "2C" CHECK:	4,217 Hours (Sept 2003)	Structural Inspection (SI) Tasks every 24,500 hours	
TIME SINCE "3C" CHECK:	7,135 Hours (Jan 2002)		
TIME SINCE "SI" CHECK:	9,759 Hours	Gear Interval: 10 Years or 20,000 cycles	
LANDING GEAR TSO:	NLG=15,515 cycles (Due April-2008), Left=11,887 (Due Sept-2009), Right=13,748 (Due Feb-2009)		

ENGINE STATUS

ENGINE TYPE:			JT8D-15	ENGINE MANUFACTURER:		Pratt & Whitney		
ENGINE MTX. PROGRAM:			On Condition with Pratt & Whitney Life Limits					
NOISE COMPLIANCE STATUS:			FAR Part 36 Stage II Compliant (No hushkit installed)					
POS.	SERIAL NUMBER	TOTAL TIME	TOTAL CYCLES	HSI HOURS	HSI CYCLES	AD 2003-12-07 Next Due	REMAINING TO DISC: HOURS	REMAINING TO DISC: CYCLES
1	687328	69,208	49,130	1,726	2,025	May 2008	1,404	2,893
2	708362	43,670	27,589	1,591	1,793	July 2007	15,127	5,401

Currently both engines (ESN 687328 & 708362) are serviceable spares located in Santiago

GENERAL DATA

FUEL CAPACITY:	34,572 POUNDS	INTERIOR:	Passenger configuration: 120 economy seats
	5,160 U.S. GALLONS		Four Galleys – two forward and two aft, two lavatories – one
FUEL SYSTEM:	Standard – Three Tank System		forward and one aft. B/E Aerospace PTC990 passenger seats.
FQIS / FF SYSTEM:	Type: Analog Units: Pounds	APU:	GTCP85-129 – S/N P34941

JT8D-15A
LIFE LIMITED PART SUMMARY
S/N 687328

Total Engine Hours:	69208	**TSHSI:**	1726
Total Engine Cycles:	49130	**CSHSI:**	2025
As of:	6/21/06		

Stage	Part Number	Serial Number	Total Hours	Total Cycles	Hour Limit	Cycle Limit	Hours Remaining	Cycles Remaining
Low Pressure Compressor								
C1 DISK	817401	BBDUAJ6181	15781	13565	N/A	20000	N/A	6435
C2 DISK	5002402-01	BDUAM9346	12040	7790	30000	20000	17960	12210
C3 DISK	800803	BBDUAJ9417	13149	8293	30000	20000	16851	11707
C4 DISK	799504	BBDUAT8649	7767	5971	30000	20000	22233	14029
C5 DISK	745705	BBDUAM9715	13149	8293	30000	20000	16851	11707
C6 DISK	745706	BBDUAN1914	13149	8293	30000	20000	16851	11707
High Pressure Compressor								
C7 DISK	774407	T04588	23411	13360	30000	17000	6589	3640
C8 DISK	787208	S39689	23411	13360	30000	20000	6589	6640
C9 DISK	701509	T18889	23411	13360	30000	19000	6589	5640
C10 DISK	772510	BENCAL4456	13149	8293	30000	20000	16851	11707
C11 DISK	772511	BENCAN7849	7767	5971	30000	20000	22233	14029
C12 DISK	798512	BENCAP0952	7767	5971	30000	20000	22233	14029
C13 DISK	5003613-01	B113AA0164	13149	8293	30000	20000	16851	11707
High Pressure Turbine								
T1 DISK	5004301-01	P02938	28105	15550	30000	20000	1895	4450
Low Pressure Turbine								
T2 DISK	802802	BLDLB41074	7767	5971	30000	20000	22233	14029
T3 DISK	795603	T02241	23411	13360	30000	20000	6589	6640
T4 DISK	769104	P04383	28596	17107	30000	20000	1404	2893

First Hour Limit - 1404 T4 DISK
First Cycle Limit - 2893 T4 DISK
AD 2003-12-07 Limit - May 2008

Note: This engine is currently a serviceable spare located in Santiago, Chile.

JT8D-15A
LIFE LIMITED PART SUMMARY
S/N 708362

Total Engine Hours:	43670	**TSHSI:**	1591
Total Engine Cycles:	27589	**CSHSI:**	1793
As of:	6/21/06		

Stage	Part Number	Serial Number	Total Hours	Total Cycles	Hour Limit	Cycle Limit	Hours Remaining	Cycles Remaining
Low Pressure Compressor								
C1 DISK	848001	BBDUAR2505	9290	9139	N/A	20000	N/A	10861
C2 DISK	790832	BBDUAU8943	11683	11481	30000	20000	18317	8519
C3 DISK	799773	BBDUAP7033	13947	13599	30000	20000	16053	6401
C4 DISK	799504	BBDUAS5893	13947	13599	30000	20000	16053	6401
C5 DISK	745705	BBDUAT8891	13947	13599	30000	20000	16053	6401
C6 DISK	745706	BBDUAU1427	13947	13599	30000	20000	16053	6401
High Pressure Compressor								
C7 DISK	5006007-02	BENCAN5812	13947	13599	30000	20000	16053	6401
C8 DISK	787208	BENCAN7011	13947	13599	30000	20000	16053	6401
C9 DISK	701509	NENCAH2067	13947	13599	30000	19000	16053	5401
C10 DISK	772510	BENCAN8555	13947	13599	30000	20000	16053	6401
C11 DISK	772511	BENCAM7831	13947	13599	30000	20000	16053	6401
C12 DISK	798512	BENCAN8047	13947	13599	30000	20000	16053	6401
C13 DISK	5003613-01	BBDUAS8284	13947	13599	30000	20000	16053	6401
High Pressure Turbine								
T1 DISK	5004301-01	BKLBCA7472	14873	14147	30000	20000	15127	5853
Low Pressure Turbine								
T2 DISK	803122	BLDLB30652	13947	13599	30000	20000	16053	6401
T3 DISK	795603	BLDLB61041	13947	13599	30000	20000	16053	6401
T4 DISK	769104	BLDLB41836	13947	13599	30000	20000	16053	6401

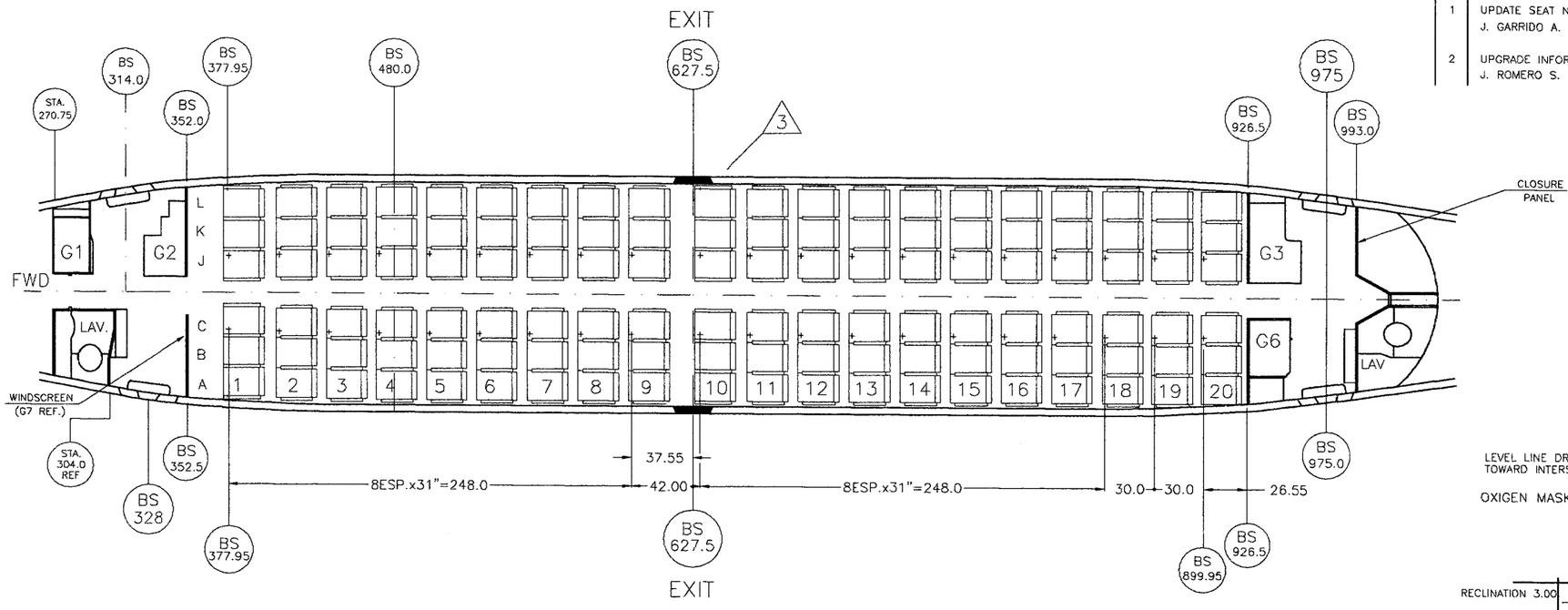
First Hour Limit - 15127 T1 DISK
First Cycle Limit - 5401 C9 DISK
AD 2003-12-07 Limit - July 2007

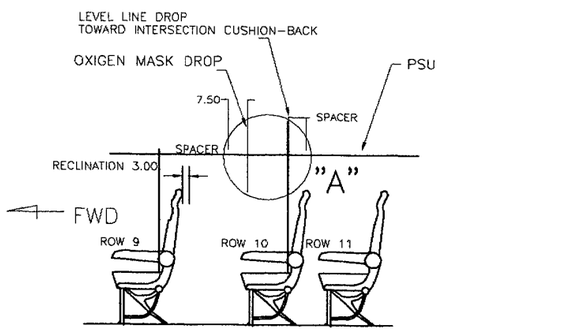
Note: This engine is currently a serviceable spare located in Santiago, Chile.

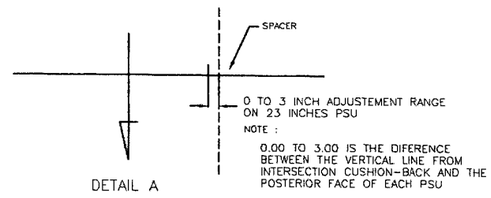
Avionics/Communications Equipment			
System	**Vendor**	**Model- P/N**	**Qty.**
Static System – Pitot	BOEING	- - - - -	2
Alternative Static System – Pitot	BOEING	- - - - -	1
Altimeter Set (Pneumatic)	KOLLSMAN	A41869-10-015	2
Air Data Computer (A.D.C.)	HONEYWELL	HG480B-50	1
Speed Indicator / MACH	SMITHS KOLLSMAN	WL104AMAJA10 A41439-10-019	2
Vertical Speed indicator	TELEDYNE	SLZ9145	2
Attitud Direction Indicador (ADI)	COLLINS	329B-8F / 522-4430-005	2
Horizontal Situation Indicator (HSI)	COLLINS	331 A-8G / 772-5370-003	2
Standby ADI	SFENA	H321AKM1	1
Compass	US GAUGE	C-5C	1
Compass Rack	SPERRY	C-6E / 614937-101	2
Clock	AEROSONIC DAVTRON	88000-4102 811B	2
Outside Air temperature (OAT)	LEAR SIEGLE	147795-01-01	1
P.D.C. Computer	- - - - - -	- - - - - -	- -
N1 Indicator	GENERAL ELECTRIC	8DJ81-WAG4 / WAG4U	2
N2 Indicator	GENERAL ELECTRIC	8DJ81-LYV4 / LYV4U	2
EGT Indicator (TGT)	US GAGE	SEL0C19DJ / SEL0C19E	2
Fuel Flow Indicator (FF)	GENERAL ELECTRIC	8DJ88-LWX3	2
Oil Quantity Indicator	SIMMONDS	393020-003 / 393020-01789	2
Oil Temperature Indicator	THE LEWIS	162BL802A	2
Oil Pressure Indicator (PSI)	US GAUGE	SRL-0C4N	2
EPR Indicator	HONEYWELL	JG298A4 / A5	2
Fuel Quantity Indicator (1 & 2)	SIMMONDS	393026-01837	2
Fuel Quantity Indicator (Central)	SIMMONDS	393026-047	1
VHF	COLLINS	618M-3 / 622-1181-001 / 618M-2D / 522-	2
HF	COLLINS	618T-2 / 522-1501-041	1
E.L.T.	CEIS TM	A06V2 / 95N6088	1
V.O.R. / I.L.S. Receiver	COLLINS	51RV-4B / 622-3257-001 // 51RV-4 / 622	3
ILS Receiver	- - - - - -	- - - - - -	- -
A.D.F. Receiver	COLLINS	51Y-4 / 522-1836-00	2
D.M.E. Interrogator	COLLINS	860E-4 / 622-2920-002	2
RMI Indicator	SPERRY	C6E / 1784460-655	2
Marker Beacon Receiver	COLLINS	51Z-4 / 522-2996-011	1
A.T.C. Transponder	COLLINS	621A-3 / 522-2703-011	2
Weather Radar	BENDIX	RDR-1E / 2067157-0153	1
G.P.W.S. Computer	SUNDSTRAND	MARK II / 965-0376-070	1
Cockpit Voice Recorder	FAIRCHILD	A100	1
Flight Data Recorder	ALLIED SIGNAL	980-4120-RQUS	1
Radio Altimeter Transceiver	COLLINS	860F-1 / 522-3698-003	1
Global Positioning System (G.P.S.)	TRIMBLE	2101 I/O APP. PLUS / 81440-02-241-D	1
Selcal Decoder	MOTOROLA	NA-135	1
Automatic Flight Guidande System			
Auto Pilot (A/P)	SPERRY	SP-77	1
Flight Director	COLLINS	FD-109	2
A/P Accessory Unit	BOEING	65-52812-52	1
Yaw Damper System	HONEYWELL	4084042-911	1
Mach Trim Coupler	SPERRY	2590650-901	1

Infromation provided by LAN

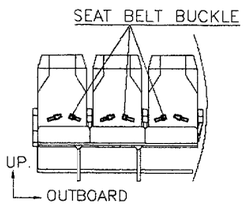
NEXT USING DRAWING:		
REVISION		
REV	DESCRIPTION	APPROVAL
1	UPDATE SEAT NUMBER IN CONFIGURATION TABLE. J. GARRIDO A. 29 NOV 2004	H. GÜNTHER W.
2	UPGRADE INFORMATION AND UPDATE EFECTIVITY J. ROMERO S. 10-ENE-2005	H. GÜNTHER W.





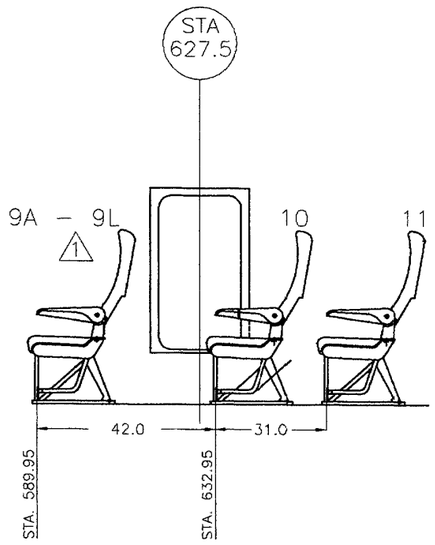


DETAIL A



CAUTION:

FOR SEAT BELT INSTALLATION CHECK THAT THE BUCKLE GOES INTO OUTBOARD SIDE ALSO IT HAS TO BE CHECKED THAT BUCKLE IS FACE UP (YOU HAVE TO SEE THE WORD LIFT) EXCEPT TO THIS OUTBOARD POSITION, WHERE BUCKLE GOES INTO INBOARD SIDE.



VIEW LOOKING OUTBOARD
EXIT ARRANGEMENT

DESIGNED		DATE
DRAFTMAN J. ROMERO S.		10-ENE-2005
CHECKED J. GARRIDO A.		10-ENE-2005
GROUP G. ALCAINO F.		10-ENE-05
APPROVAL H. GUNTHER W.		10.01.05
PROJECT : D:/RESPALDO/J.ROMERO/J.GARRIDO		
VERSION A/C: B737-200 CC-CYK		

FILE REF.: LA0426-0002.DWG
DIMENSIONS IN INCHES (MM)
BREAK ALL SHARP EDGES

ALL TOLERANCES +/- .030
HOLES +/- 0.030
ANGLES +/- 30'
UNLESS OTHERWISE SPECIFIED.

LAN

VICE PRESIDENCIA TECNICA
GERENCIA INGENIERIA
SUBGERENCIA INGENIERIA
DPTO. INGENIERIA DE PROYECTOS

CABIN ARRANGEMENT
120 PASSENGER
BOEING 737 - 200

DWG N°	REV
3-0277-00-000	2

SCALE : ---	STANDARD: ISO-A	SH	1 OF 2

NEXT USING DRAWING:		
REVISION		
REV	DESCRIPTION	APPROVAL
1	UPDATE SEAT NUMBER IN CONFIGURATION TABLE. J. GARRIDO A. 29 NOV 2004	H. GÜNTHER W.
2	UPGRADE INFORMATION AND UPDATE EFECTIVITY J. ROMERO S. 10-ENE-2005	H. GÜNTHER W.

SEAT CONFIGURATION FOR A/C CC-CYK

SEAT LOCATION ROW	MODEL N° LH. SIDE	MODEL N° RH. SIDE	SEAT DESCRIPTION	SEAT PER S/S LH/RH	PASS PER S/S
(1)	990-UC102M-3L	990-UC102M-3R	IN ARM TABLE; NARROW	1	3
(2)	990-UC102A-3L	990-UC102A-3R	NARROW	1	3
(3)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(4)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(5)	990-UC102A-3L	990-UC102C-3R	STANDARD	1	3
(6)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(7)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(8)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(9)	990-UC102G-3L	990-UC102G-3R	SPECIAL; NO RECLINE; NO REAR FOOD TRAY TABLE	1	3
(10)	990-UC102E-3L	990-UC102E-3R	IN ARM TABLE; NO BREAK OVER	1	3
(11)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(12)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(13)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(14)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(15)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(16)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(17)	990-UC102A-3L	990-UC102A-3R	STANDARD	1	3
(18)	990-UC102F-3L	990-UC102F-3R	STANDARD	1	3
(19)	990-UC102C-3L	990-UC102C-3R	NARROW	1	3
(20)	990-UC102D-3L	990-UC102D-3R	NARROW; NO REAR FOOD TRAY TABLE	1	3
			TOTAL	40	120

GENERAL NOTES:

1. PER USA 14 CFR, 25.813 (c) (1), SEAT RECLINE MUST BE INOPERATIVE FOR SEATS 9 ABC AND 9 JKL.
2. ALL SEATS MUST COMPLY WITH TSO C39B, NAS809, FAR 25.853 AND CONTAIN FAA APPROVED UNDER SEAT BAGGAGE RESTRAINT BARS
3. OUTBOARD SEAT BACK BREAKOVER MUST BE LIMITED TO ENSURE THAT IT WILL NOT BLOCK THE EMERGENCY EXIT OPENING. OUTBOARD ARMREST MUST BE INSTALLED ON THE TYPE III EXITS.
4. AISLE WIDTH MAIN AISLE MUST NOT BE LESS THAN 15" BELOW 25" HEIGHT, AND 20" ABOVE 25" HEIGHT
5. AISLE WIDTH LEADING TO ALL TYPE I AND TYPE II EXISTS MUST NOT BE LESS THAN 20" PLUS ASSIST SPACE PER USA CFR 25.813 (b)
6. ALL SEAT LOCATIONS ARE TO CENTERLINE OF FORWARD SEAT STUD WITH A TOLERANCE OF +/- 3/8"
7. MINIMUM DISTANCE FROM BULKHEAD TO FORWARD FACE OF SEAT DIRECTLY AFT. OF BULKHEAD IS 35"
8. AIRCRAFT WEIGHT AND BALANCE MUST BE RECALCULATED (OR AIRCRAFT WEIGHED) AFTER INTERIOR INSTALLATION TO DETERMINE NEW AIRCRAFT WEIGHT AND C.G. LOCATION . REVISE AIRCRAFT WEIGHT AND BALANCE RECORD AS REQUIRED.
9. SEATS MUST NOT OBSTRUCT/BLOCK TYPE III EXIT OPENING.
10. RE-PITCH PSUs AS REQUIRED TO MATCH NEW SEAT POSITIONS
11. ACCOMPLISH DROP CHECK OF AIRCRAFT PASSENGER OXYGEN SYSTEM AFTER AIRCRAFT RECONFIGURATION PER M.M. CHAPTER 35.
12. EMERGENCY EQUIPMENT TO BE INSTALLED IN ACCORDANCE WITH SEPARATE APPROVED DRAWING.
13. ALL ITEMS NOT CALLED OUT ON THIS DRAWING ARE EXISTING ON AIRCRAFT OR ARE INSTALLED IN ACCORDANCE WITH OTHER APPROVED DRAWINGS.

FILE REF.: LA0426-0102.DWG
DIMENSIONS IN INCHES (MM)
BREAK ALL SHARP EDGES

ALL TOLERANCES +/- .030
HOLES +/- 0.030
ANGLES +/- 30'
UNLESS OTHERWISE SPECIFIED.

DESIGNED		DATE
DRAFTMAN J. ROMERO S.		10-ENE-2005
CHECKED J. GARRIDO A.		10-ENE-2005
GROUP G. ALCAINO F.		10 ENE 2005
APPROVAL H. GUNTHER W.		10.01.05
PROJECT : D:/RESPALDO/J.ROMERO/J.GARRIDO		
VERSION A/C: B737-200 CC-CYK		

LAN

VICE PRESIDENCIA TECNICA
GERENCIA INGENIERIA
SUBGERENCIA INGENIERIA
DPTO. INGENIERIA DE PROYECTOS

CABIN ARRANGEMENT
120 PASSENGER
BOEING 737 - 200

DWG N° 3-0277-00-000 REV 2

SCALE : --- | STANDARD: ISO-A | SH | 2 OF 2

Airframe Airworthiness Directive Repetitive / Open Inspections

Aircraft B737-205 ADV CC-CYK MSN 21445 LN 506 VN PJ006 **TAT: 65721** **TAC: 76639** **as of: 6/21/2006**

AD No.	AD Subject	Last Accomplished Date	Hours	Cycles	Interval		Next Due	Time Remaining	
74-08-09 R2	Lavatory fire prevention	2/2/2006	65121	75930	1000	H	66121	400	Hours
87-08-09	Inflate tires with nitrogen placards	Incorporated into the maintenance program & accomp on J/C PT-03							
88-22-09	Operational & functional check of takeoff warning system	3/21/2006	65354	76205	600	H	65954	233	Hours
	***Note:** The AD mandated interval for the operational and functional check is 200 hours. The data provided indicates accomplishment of the requirements of this AD is per an Alternate Means of Compliance.*								
88-22-11	HFEC and detailed visual inspectionalong the lap joints and str. 17	4/12/2006	65411	76277	450	D	7/6/2007	380	Days
		4/12/2006	65411	76277	4500	C	80777	4138	Cycles
	Detailed visual inspection of the tearstraps	4/12/2006	65411	76277	1440	D	3/22/2010	1370	Days
		4/12/2006	65411	76277	12000	C	88277	11638	Cycles
89-11-06 R1	Inspect for cracks at the forward lower cargo compartment frames			71760	9000	C	80760	4121	Cycles
89-14-11	Inspect the flight attendants seat pan roller fittings	10/29/2005	64648	75377	545	D	4/27/2007	310	Days
90-12-11 R1	Inspect for frayed or broken escape slide release cables	10/10/2005	64572	75278	365	D	10/10/2006	111	Days
90-25-01	Aging Airplane Corrosion Prevention and Control Program	Program current per LAN - awaiting status							
91-07-04	Ultrasonic inspection for delamination of the window belt skin doubler	3/22/2005	63950	74561	6000	C	80561	3922	Cycles
92-09-02	Inspect fuel feed lines	6/1/2005	64281	74940	7000	H	71281	5560	Hours
	***Note:** Inspection to be repeated every 2nd "C" check or 7,000 flight hours, W.O.F.*								
92-15-03	Inspect main landing gear brakes	Incorporated into Lan Chile's maintenance program							
92-25-09	Visual/HFEC inspections for delamination to prevent decompression	4/12/2006	65411	76277	450	D	7/6/2007	380	Days
		4/12/2006	65411	76277	4500	C	80777	4138	Cycles
93-08-04	Aging Airplane Structural Inspection Program								
	SB 53-1066R3 - Stringer 18A shear beam inspect/repair/modify			74940	4000	C	78940	2301	Cycles
	SB 53-1085R1 - Fuselage stringer-to-frame tie clip inspect/replace	11/3/2000	55963	65995	15000	C	80995	4356	Cycles
93-14-10	Insp forward/aft body frames adjacent to aft lower cargo door	11/3/2000	55963	65995	28000	C	93995	17356	Cycles
95-12-17	Inspect O/B chord of frame at BS 727 and in O/B chord of stringer 18A	2/4/2002	58583	68683	30000	C	98683	22044	Cycles
	***Note:** Data provided indicates that accomplishment of the replacement of the outboard chord per AD par. (f) is for RH side only.*								
97-14-04	Leak test of the main rudder PCU	6/8/2005	64281	74940	6400	H	70681	4960	Hours
98-11-04 R1	Supplemental Structural Inspection Documents	Refer to attached status							
98-14-09	Inspect the lower flange of FEMS fitting for fatigue cracking	3/20/2006	65348	76197	700	C	76897	258	Cycles
98-22-10	Inspect the frame support structure of the forward service doorway	7/6/2005	64281	74940	4500	C	79440	2801	Cycles
99-08-23	Inspect the aft pressure bulkhead	2/28/2006	65264	76093	1200	C	77293	654	Cycles
99-10-12	Inspect the LH / RH MLG actuator beam arm clevis	5/29/2006	65612	76515	90	D	8/27/2006	67	Days
99-11-05	Perform a displacement test of the secondary slide in the dual servo valve in the rudder PCU		63269		24000	H	87269	21548	Hours

Airframe Airworthiness Directive Repetitive / Open Inspections

Aircraft B737-205 ADV CC-CYK MSN 21445 LN 506 VN PJ006 **TAT: 65721** **TAC: 76639** **as of: 6/21/2006**

AD No.	AD Subject	Last Accomplished Date	Hours	Cycles	Interval		Next Due	Time Remaining	
99-21-15	Inspect the fuel boost pump wiring for damage		50035		30000	H	80035	14314	Hours
99-24-08	Perform diode test - par. (c)	5/31/2006	65625	76530	600	H	66225	504	Hours
	Replace airplane battery with new or reconditioned battery - par. (f)	5/31/2006	65625	76530	750	H	66375	654	Hours
99-26-07	Inspect the aft mount cone bolt indicator for proper alignment, par. (a) (1)		Accomplished by Lan Chile at every terminal check						
2000-05-29	Inspect the forward pressure bulkhead for fatigue cracking	1/17/2005	63597	74169	6000	C	80169	3530	Cycles
2000-25-07	Inspect inboard flap tracks of the LH wing outboard flap	2/26/2006	65252	76078	1200	C	77278	639	Cycles
	Inspect inboard flap tracks of the RH wing outboard flap	2/25/2006	65246	76071	1200	C	77271	632	Cycles
2001-17-18	Main wheel pressure floor transverse beams and floor beams insp	6/21/2005	64281	74940	6000	C	80940	4301	Cycles
2002-01-01	Inspect the elevator tab assembly on the LH & RH sides of the	4/22/2006	65442	76316	2000	H	67442	1721	Hours
	airplane IAW WP II of SB 737-55A1070	4/22/2006	65442	76316	1500	C	77816	1177	Cycles
	Free play inspections of the elevator tab assembly on the LH & RH	10/17/2004	63161	73671	6000	H	69161	3440	Hours
	sides of the airplane IAW WP III of SB 737-55A1070	10/17/2004	63161	73671	4500	C	78171	1532	Cycles
		10/20/2004	63179	73686	6000	H	69179	3458	Hours
		10/20/2004	63179	73686	4500	C	78186	1547	Cycles
	***Note:** This AD has been superseded by AD 2006-12-23, effective date July 3, 2006; however, Lan Chile states that AD 2006-12-23 is "in analysis".*								
2002-07-08	LFEC insp to find cracking of the lap joints of the fuselage - par. (i)	6/18/2005	64281	74940	5000	C	79940	3301	Cycles
2002-10-11	Inspect the aft pressure bulkhead at BS 1016	6/16/2005	64281	74940	730	D	6/16/2007	360	Days
2003-14-06	External inspection of the fuselage skin on lap joints at BS 259 to 1016	6/22/2006	65721	76639	500	C	77139	500	Cycles
2003-24-08	NDT inspect each carriage spindle of the left / right outboard mid-flaps,	5/5/2006	65507	76395	1500	C	77895	1256	Cycles
	para (f) IAW SB 737-57A1277 Work Package I	5/5/2006	65507	76395	60	D	7/4/2006	13	Days
2004-06-18	Conduct leak tests of the toilet tank dump valve and the service panel	1/3/2006	64971	75763	1000	H	65971	250	Hours
	drain valve and inspect the outer/cap door and seal mating surface	1/3/2006	64971	75763	180	D	7/2/2006	11	Days
		7/14/2005	64281	74940	5000	H	69281	3560	Hours
2004-18-06	Perform an external detailed and eddy current insp of the crown area and other known areas of fuselage skin cracking	4/12/2006	65411	76277	4500	C	80777	4138	Cycles
	Perform an external detailed insp of the lower lobe area and section 41 of the fuselage for cracking	4/12/2006	65411	76277	9000	C	85277	8638	Cycles
2004-19-10	Perform a detailed insp & magnetic particle insp for corrosion & cracking	6/20/2005	64281	74940	1460	D	6/19/2009	1094	Days
	of the horizontal stabilizer hinge pins IAW Part 2 or 3 of SB 55A1077	6/20/2005	64281	74940	6000	H	70281	4560	Hours
2005-07-12	Insp for cracks arounf the frame web around the cutout for the doorstop intercostal strap at the aft side of the B.S. 291.5 at stringer 16L	4/7/2006	65411	76277	4500	C	80777	4138	Cycles
2005-20-01	Inspect the stiffeners at LBL and RBL 6.15 on the rear spar pf the wing center section for cracks	4/22/2006	65442	76316	4500	C	80816	4177	Cycles

Airframe Airworthiness Directive Repetitive / Open Inspections

Aircraft B737-205 ADV CC-CYK MSN 21445 LN 506 VN PJ006 **TAT: 65721** **TAC: 76639** **as of: 6/21/2006**

Open Airworthiness Directives		Threshold	Due at		Time Remaining	
86-12-04	Inspect the horizontal stabilizer center section rear spar upper chord	50000 cyc after chord replacement	100995	TAC	24356	Cycles
86-12-05	Inspect the horiz stabilizer ctr section rear spar upper chord attach lug	40000 cyc after chord replacement/mod	93521	TAC	16882	Cycles
90-06-02	Aging Airplane Structural Modification Program	Refer to the Aging Airplane SMP section of this report for details				
93-17-08	Aging Airplane Structural Modification Program	Refer to the Aging Airplane SMP section of this report for details				
2000-06-13 R1	Modify the four corners of the door frame and the cross beams of the aft cargo door	12000 cyc from eff. date of AD	79834	TAC	3195	Cycles
2002-05-07	Insp the upper flange of I/B track of each O/B flap at rear spar attach	10 yrs after rework	10/25/2010		1587	Days
2002-07-08	LFEC inspection to find cracking of the lap joint repair - par. (m)	45000 cyc after repair	110995	TAC	34356	Cycles
2002-20-07 R1	Install a new rudder control system	6 yrs from eff. date of AD	11/12/2008		875	Days
	***Note:** Accomplishment of the actions required per paragraph (a) of this AD constitutes terminating action for the requirements of AD 97-14-04.*					
2003-07-12	Examine records and/or landing gear parts to determine cycles on parts	10 yrs from eff. date of AD	5/20/2013		2525	Days
2003-24-08	Gap check of I/B & O/B carriage of the left and right outboard mid-flaps, para (d) IAW SB 737-57A1277 Work Package II	3070 cyc on OH carriage spindles	78014	TAC	1375	Cycles
2004-03-34 R1	Replace scape slide latch assembly mounting fastener	3 yrs from eff. date of AD	7/21/2007		395	Days
2004-06-18	Conduct leak tests of the toilet tank dump valve and the service panel drain valve and inspect the outer/cap door and seal mating surface	5000 hrs from eff. date of AD	67426	TAT	1705	Hours
		4500 hrs from eff. date of AD	66926	TAT	1205	Hours
2004-19-10	Perform a detailed insp & magnetic particle insp for corrosion & cracking of the horizontal stabilizer hinge pins IAW Part 3 of SB 55A1077	12000 hrs or,	75247	TAT	9526	Hours
		96 mths from eff. date of AD, WOF	10/17/12		2310	Days
2005-04-01	Perform an electrical bondingresistance test between the bulkhead fitting of the engine fuel feed tube and the front spar inside fuel tank of wings	5 yrs from eff. date of AD	3/23/10		1371	Days
2005-07-19	Insp fuselage skin, doubler, bearstrap, and frames around main forward cargo door and aft cargo door I.A.W SB 737-53A1228					
	Skin & Bearstrap Inspections On Cargo Door Cutouts - Table 1	4000 cyc from eff. date of AD	78827	TAC	2188	Cycles
	Forward Cargo Doorway Frame Inspections - Table 2	4000 cyc from eff. date of AD	78827	TAC	2188	Cycles
	Aft Cargo Doorway Frame Inspections - Table 3	4000 cyc from eff. date of AD	78827	TAC	2188	Cycles
	***Note:** This AD requires compliance within the specified compliance time after the effective date (May 12, 2005) of this AD. The SB mandates a compliance time after the SB date (July 10, 2003). It appears as though Lan Chile is tracking the initial compliance due from the less advantageous SB date and not the AD effective date. Further review of the aircraft documentation would be necessitated to determine the actual due times for this AD.*					
2005-13-15	One time inspection of the secondary fuel vapor barrier of the wing center section	4 yrs from eff. date of AD	7/29/2009		1134	Days
2005-13-30	Inspect the fuselage side skins just aft of the main wheelwell	4500 cyc from eff. date of AD	79476	TAC	2837	Cycles
2005-18-08	Inspect the retaining pin lugs of the support fitting of the MLG beam	3000 cyc from eff. date of AD	78286	TAC	1647	Cycles
2005-19-25	Inspect for cracks arounf the satellite holes of the radio altimeter cutout	4500 cyc from eff. date of AD	79897	TAC	3258	Cycles
2005-20-03	Inspect the intercostal web, attachment clips and stringer splice chanels on the forward and aft sides of the forward entry door for cracks	4500 cyc from eff. date of AD	79898	TAC	3259	Cycles

Airframe Airworthiness Directive Repetitive / Open Inspections

Aircraft B737-205 ADV CC-CYK MSN 21445 LN 506 VN PJ006 **TAT: 65721** **TAC: 76639** **as of: 6/21/2006**

Open Airworthiness Directives		Threshold	Due at		Time Remaining	
2005-20-16	Perform a one time DVI of the I/B attach fittings at slats 2 & 5 of the wing L/E slat tracks	12 mths from eff. date of AD	11/9/06		141	Days
	Replace the aluminum I/B attach fittings with new, improved steel fittings	30000 cyc from eff. date of AD, or	105449	TAC	28810	Cycles
		120 mths from eff. date of AD, WOF	11/9/15		3428	Days
2005-23-17	Inspect to correct chafing of the wiring behind the P15 refuel panel	18 mths from eff. date of AD	6/21/07		365	Days
2005-26-03	Modify the elevator input torque tube assembly	60 mths from eff. date of AD	1/20/11		1674	Days
2006-03-12	Insp. to detect chaffing of the wire bundles located below the passenger compartment, above the center fuel tank, aft of sta 540	60 mths from eff. date of AD	3/16/11		1729	Days
2006-07-12	Insp. for scribe lines and cracks in the fuselage skin - SB 737-53A1262	80899 TAC	80899	TAC	4260	Cycles
2006-12-23	Insp the elevator tab assembly on the left & right sides of the airplane	*Refer to AD 2002-01-01 for more information / current status*				
	***Note:** This AD became effective on July 3, 2006, and has superseded AD 2002-01-01. Lan Chile lists this AD within their AD Status Airframe status report as " AD in Analysis" and also is currently still tracking the requirements of AD 2002-01-01 and has not yet to incorporate the new AD into their status reports.*					
2006-13-13	Revise the AFM Cabin Pressurization and Emergency Procedures sect.	60 dys from eff. date of AD	9/7/2006		78	Days
	***Note:** Lan Chile includes this AD within their "AD Status Airframe" report, dated July 5, 2006. This AD was issued on July 7,2006 and as such, it current status indicates more days remaining than the interval.*					

***NOTE:** This AD summary was generated for records review purposes only and is not an official compliance record to determine airworthiness. This AD listing contains the status of applicable open and repetitive FAA ADs that have been issued thru bi-weekly 2006-13 - released June 26, 2006, as reported by Lan Chile. A detailed audit would be required to verify the accuracy of the information contained herein.*

AGING AIRPLANE
STRUCTURAL MODIFICATION PROGRAM AD 90-06-02 AND 93-17-08
B737-205 ADV
REG. No. CC-CYK S/N 21445 L/N 506 Variable No. PJ006

Effective: June 21, 2006 TAT: 65,721 TAC: 76,639

Service Bulletin	Subject	Threshold	Modification Status
AD 90-06-02			
32A1113 R4	Main landing gear torsion link pins inspection and replacement	Modify by 12-31-92	Verified PCW on 2-5-99 per Messier WO 141150.1.1 & on 9-30-99 per Messier WO 142712.1.1
32-1123	Main landing gear lubrication provisions and fuse bolt replacement	Modify by 12-31-92	Terminated on 8-2-99 per Messier service WO 141150.1.1
52A1038 R4	Main cargo door skin inspection and doubler installation	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
52-1059 R2	E & E compartment door inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
52-1064 R2	Electronic equipment access door latch pin replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
52-1065 R6	Forward and aft cargo door stop fittings inspection and replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 7-10-05 per EO AD-52-0063
52-1068 R2	Main cargo door frame reinforcement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
52-1079 R5	Door – aft cargo – frame and beam inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 7-14-05 per EO AD-52-0010R5
52-1090 R1	Doors – forward cargo door forward and aft frames – inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 7-14-05 per EO AD-52-0045R1
52-1100 R1	Doors – forward cargo – lower forward corner inspection, modification, and replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 2-4-02 per EO AD-52-0027
53-1023 R11	Control cabin “E-F” window post inspection and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
53A1027 R7	Forward and aft cargo compartment body frame inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 9-14-04 per EO AD-53-0205

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Service Bulletin	Subject	Threshold	Modification Status
53-1031 R4	Wing to body drag angle inspection and replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
53A1042 R9	Lower lobe hot bonded skin panel inspection, repair, and panel replacement	Modify by 75,000 cycles or 20 years of airframe age, whichever occurs first, or by 4-17-94, whichever occurs later	Not applicable
53-1051 R5	Forward lower cargo doorway frame inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 7-5-05 per EO AD53-0470R1
53-1054 R2	Nose wheel well forward bulkhead reinforcement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
53-1058 R3	Forward airstair stowage doorway inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 12-18-99 per EO AD-53-0194
53-1061 R4	Overwing frame inspection, modification, and repair – Body stations 559 through 639	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 12-16-99 per EO AD-53-0195
53-1062 R2	Body station 1138 frame shear tie inspection and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 11-25-99 per EO AD-53-0196
53A1064 R6	Body frame inspection and modification at forward airstair doorway sill – body stations 351.2 and 360	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 8-25-82 per EO TO 60-53-50. Last action on 3-15-05 per EO AD-53-0070
53-1065	Fuselage – aft skin – inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Modification requirement deleted per Boeing Document D6-38505 Rev. G.
53-1067	Inspection and fastener additions – Stringer 18A to body station 663.75 attachment area	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 2-19-86 per EO TO 60-57-81
53-1069 R1	Fuselage – forward cargo compartment cargo damage prevention	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 12-16-99 per EO AD-53-0197
53-1074	Fuselage – entry and galley doorway skin cut-outs – inspection and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Modification requirement deleted per Boeing Document D6-38505 Rev. F.
53-1076 R4	Fuselage – circumferential butt splices and bonded doublers – inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 11-3-00 per EO AD-53-0200R1

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Service Bulletin	Subject	Threshold	Modification Status
53-1077 R2	Fuselage skin inspection, repair, and preventative modification aft of body station 727, between S-18 and S-21	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
53-1083 R3	Fuselage – forward entry doorway structure at lower sill – inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 6-16-05 per EO AD-53-0001R1
53-1086 R3	Fuselage – body station 1156 outboard chord inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 2-8-93 per EO TO 60-53-0164 and on 3-10-05 per EO AD-53-0311/-0312
53-1096 R5	Fuselage – aft lower cargo doorway frame inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 10-3-00 per EO AD-53-0071R1
53-1099 R1	Fuselage – body station 540 bulkhead forging inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
53A1108 R5	Fuselage – forward service doorway aft frame inspection and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 7-7-05 per EO AD-53-0075R1
53-1116 R3	Fuselage – forward galley doorway upper corners – inspection, repair, and preventative modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 10-14-00 & 3-10-05 per EO AD-53-0204
55A1020 R4	Elevator rear spar inspection and modification or repair at tab inboard hinge attachment	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
55-1022 R1	Elevator rear spar inspection and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
55-1026 R3	Horizontal and vertical stabilizer rear spar attach bolt replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 9-8-95 per EO 2-55-0894. Last action on 3-10-05 per EO AD-55-0008
55-1028 R8	Horizontal stabilizer and vertical fin attachment and lug inspection and modification	Modify by 20 years of airframe age or by 4-17-94, whichever occurs later	Terminated on 9-8-95 per EO 2-55-0895R1
55A1029 R4	Horizontal stabilizer center section rear spar attach lug inspection	Modify by 20 years of airframe age or by 4-17-94, whichever occurs later	Not applicable

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Service Bulletin	Subject	Threshold	Modification Status
55A1031 R4	Horizontal stabilizer center section rear spar upper chord inspection, modification, repair and replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
55-1033 R1	Horizontal stabilizer center section rear spar upper chord attach lug inspection and modification	Modify by 75,000 cycles or 20 years of airframe age, whichever occurs first, or by 4-17-94, whichever occurs later	Terminated on 9-8-95 per EO 2-55-0930R1
55-1034 R2	Horizontal stabilizer center section rear spar upper chord inspection, modification, repair, and replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 11-9-00 per EO AD-55-0026
57-1045 R5	Wing center section – floor beam attachment inspection, repair, and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1071 R8	Inspection and modification of main landing gear trunnion support beam and body station 706 frame lug	Modify by 37,500 cycles or by 4-17-94, whichever occurs later	Not applicable
57A1079 R4	Wings – flap track support fitting attach bolts inspection	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1080 R3	Slat track attach fitting inspection	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 9-8-83 per EO TO 60-57-40
57A1081 R12	Wing front spar upper chord inspection / modification – front spar station 90 to 225	Modify by 4-17-94	Not applicable
57A1082 R8	Outboard trailing edge flap inboard track inspection	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1083 R2	Main wheel well pressure web to floor beam fastener replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1084 R6	Outboard trailing edge flap outboard track inspection	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1087 R9	Body buttock line 70.85 rib upr chord and BS 663.75 bulkhead ftg & skin insp, mod, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable

AGING AIRPLANE
STRUCTURAL MODIFICATION PROGRAM AD 90-06-02 AND 93-17-08
B737-205 ADV
REG. No. CC-CYK S/N 21445 L/N 506 Variable No. PJ006

Effective: June 21, 2006 TAT: 65,721 TAC: 76,639

Service Bulletin	Subject	Threshold	Modification Status
57-1099 R1	Wing center section stringer 13C to floor beam attachment modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Not applicable
57-1107 R4	Body buttock line 70.85 upper chord to stringer gap inspection and re-work	Modify by 75,000 cycles or within the next 12,000 cycles following 4-17-90, whichever occurs first, or by 4-17-94, whichever occurs later	Terminated on 11-30-84 per EO TO 60-57-58
57A1130 R3	Inboard flap outboard track aft support attachment inspection and modification	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 8-23-82 per EO TO 60-57-75R1
57-1137 R6	Body buttock line 70.85 rib upper chord and body station 663.75 bulkhead fitting and skin inspection, modification, and repair	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 2-19-86 per EO TO 60-57-81
57-1170 R3	Wings – main frame – landing gear beam swing link attachment bolt inspection or replacement	Modify by 75,000 cycles or by 4-17-94, whichever occurs later	Terminated on 12-27-89 per EO TO 60-57-91
AD 93-17-08			
32A1224 R4	Landing gear – main gear and doors – main gear trunnion pin, actuator bolt assemblies re-work	Modify by 1-31-96	Terminated LH on 10-12-96 per EO AD-53-0008 and verified per Messier WO 142712.1.1. RH verified on 2-5-99 per Messier WO 141150.1.1
52A1038 R4	Doors – cargo – main cargo door skin inspection and doubler installation	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Not applicable
53-1080 R2	Fuselage – body station 695 frame lower fitting inspection, repair, and replacement	Modify by 75,000 cycles or 20 years of airframe age, whichever occurs first, or by 10-1-97, whichever occurs later	Terminated LH on 10-12-96 per EO AD-53-0008 and RH on 9-8-95 per EO P53-0019
53-1110 R1	Fuselage – body station 663.75 frame at stringer 17 inspection, repair, and modification	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Terminated on 11-4-00 & 3-10-05 per EO AD-53-0201
53-1121 R1	Fuselage – aft lower cargo doorway frame inspection, preventative change, and repair	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Not applicable

AGING AIRPLANE
STRUCTURAL MODIFICATION PROGRAM AD 90-06-02 AND 93-17-08
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Effective: June 21, 2006 TAT: 65,721 TAC: 76,639

Service Bulletin	Subject	Threshold	Modification Status
57A1079 R4	Wings – outboard flaps – flap track support fitting attach bolts inspection and replacement	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Not applicable
57-1139 R4	Wing – main wheel well pressure floor transverse beams and floor beams inspection and mod.	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Terminated on 7-12-05 per EO AD-57-0141R1
57A1206 R1	Wings – trailing edge outboard flaps – flap track forward support fitting attach bolts inspection and replacement	Modify by 75,000 cycles or by 10-1-97, whichever occurs later	Terminated on 1-30-99 per EO TO 60-57-099

This report lists all service bulletins currently required by Airworthiness Directive 90-06-02, amendment 39-6489 and Airworthiness Directive 93-17-08, amendment 39-8679.